CAN MODELING OF REACTIVE FLOWS DESCRIBE REALITY?

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Objectives (and an Outline)

Introduce a selection of combustion configurations that we need to be able to technically improve on during the next decade.

Engineering focus.

 \Rightarrow Predictive simulations.

Summarize the modeling used in simulating turbulent combustion. Extremely challenging physics (case dependent)

Discuss VVUQ issues in combustion simulations.

- Grid refinement
- Sub-models, ...
- IC & BC issues

Present examples of (state-of-the-art?) *engineering* combustion simulations.

Try to address/discuss the usefulness of such simulations, and if they can be trusted and/or used to improve the design and/or feed the development of new (more appropriate) models.



Gas Turbine Combustors

LM Gas Turbine Combustors Aero Gas Turbine Combustors

Courtesy of A. Lindholm, Siemens, Finspång, Sweden



Scramjet Combustion Engines

Man always wants to travel faster

Aerospaceplanes, space launchers, r



RB/TBCC Scramjets & Dual-mode Scramjets

Since the mid-50's research has beer carried out on ram-/scramjet engines

Issues with supersonic injection, mixin self-ignition and flame stabilization, ... $(s_u \approx 2 \text{ m/s s}_t \approx 20 \text{ m/s in a 1800 m/s flow!})$

Ground based experimental facilities not sufficient! – Run times, conditions

(High-fidelity) simulations?





Other Important Areas of Combustion

Astrophysical Combustion

- Birth and death of the universe
- Nuclear reactions
- EOS: Completely degenerate electron gas
- It takes about 2s to blow up a White Dwarf star the size of the earth having a mass of $1.4 M_{\odot}$
- Comparison: Shape of light curve luminosity

Condensed Phase & Accidental Explosions

- Safety and weapons design as well as protection
- Often multi-phase (gas, liquid & particles)
- Comparison: Pressure-data

Internal Combustion Engines

- Transport is being more & more important to our way of life
- Reduce pollutants & increase fuel efficiency
- Spray combustion
- Comparison: Laser-based methods







Why Bother Addressing such Complex Flows?

Combustion is extremely complicated and diverse ...

- Chemical kinetics,
- Mixing, self-ignition, instabilities, near wall flows, conjugate heat transfer, acoustics, ...

Important to our way of life ...

- Where would we have been without fire (and the IC engine)?

Survival of the earth ...

- Reduce pollutants but still produce energy
- Alternatives: Wind, hydro, wave, sun, ... but often with large technical problems (storage)
- Need to continue study combustion focusing on alternative fuels

Other drivers

- We always want to travel faster ... Economical drivers ... Military drivers (spin-off)...

Is there any hope that we can predict combustion phenomena?

- YES! But it is not simple
- Combine experiments and simulations better stop fight for funds
 - Non-intrusive measurement techniques
 - Supercomputing capacity available to more research groups
- Combine fundamental research with applied research
- Persistence



Experiments (vs. Simulations)

Advanced laser and optical techniques have enabled non-intrusive measurements in at least laboratory flames (Barlow, Aldén, Wolfrum, Grisch, ...

Gives the 'true' picture of what happens in a flame

- Difficult, expensive
- Pointwise, lines & arcs
- Tomographical reconstruction







Mathematical Modeling of Combustion

Multi-physics turbulent flows is a multi-scale phenomenon with key subprocesses interacting on a wide range of length and time scales



Combustion Simulation Framework





Simulation of Non-Reactive Flows

Flow Modeling A range of flow modeling methods with different built-in features and capabilities are available



Validation

Very important to validate any RANS, LES, ..., DNS models, methods & codes





Numerical Methods (OpenFOAM)

Unstructured Finite Volume (FV) discretization

Reynolds transport (or Gauss) theorem $\overline{\mathbf{u}} = [\overline{\rho}, \overline{\rho} \tilde{Y}_i, \overline{\rho} \tilde{\mathbf{v}}, \overline{\rho} \tilde{E}]^T$

 $\partial_t(\overline{\mathbf{u}}_P) + \frac{1}{\delta V_P} \sum_f \left[F_f^C(\overline{\mathbf{u}}) - F_f^D(\overline{\mathbf{u}}) + F_f^B(\overline{\mathbf{u}}) \right] = -(\nabla p)_P + s_P(\overline{\mathbf{u}})$



Semi-Implicit Algorithm

Monotone or monotonicity-preserving reconstruction of convective fluxes Central difference approximations of inner derivatives in other fluxes Crank Nicholson time integration, Co≈0.5

Fully Explicit Algorithm

Monotone or monotonicity-preserving reconstruction of convective fluxes Central difference approximations for inner derivatives other fluxes RK time integration, Co≈0.5

Modified Equations Analysis (MEA)

Taylor series expansion used to evaluate the leading order TE $T \approx \overline{\rho}([C(\nabla \tilde{v})^{T} + (\nabla \tilde{v})C^{T} + \chi^{2}(\nabla \tilde{v})d \otimes (\nabla \tilde{v})d] + \frac{1}{8}[\tilde{v} \otimes ((\nabla^{2} \tilde{v})(d \otimes d)) + ((\nabla^{2} \tilde{v})(d \otimes d)) \otimes \tilde{v}])$

Drikakis D., Fureby C., Grinstein F.F. & Liefendahl M.; 2007, "ILES with Limiting Algorithms", In Implicit Large Eddy Simulation: Computing Turbulent Fluid Dynamics, Eds. Grinstein F.F., Margolin L. & Rider B., Cambridge University Press, p 94.



Combustion Modeling using LES

Balance equations of mass, momentum and energy for a mixture

$$G*\begin{cases} \partial_{t}(\overline{\rho})+\nabla\cdot(\overline{\rho}\tilde{\mathbf{v}})=0\\ \partial_{t}(\overline{\rho}\tilde{\mathbf{v}})+\nabla\cdot(\overline{\rho}\tilde{\mathbf{v}}\otimes\tilde{\mathbf{v}})=-\nabla\overline{p}+\nabla\cdot(\overline{\mathbf{S}}-\mathbf{B})+\overline{\rho}\tilde{\mathbf{f}}\\ \partial_{t}(\overline{\rho}\tilde{E})+\nabla\cdot(\overline{\rho}\tilde{\mathbf{v}}\tilde{E})=\nabla\cdot(-\overline{p}\tilde{\mathbf{v}}+\tilde{\mathbf{S}}\tilde{\mathbf{v}}+\overline{\mathbf{h}}-\mathbf{b}_{E})+\overline{\rho}\tilde{\sigma}\\ \partial_{t}(\overline{\rho}\tilde{Y}_{i})+\nabla\cdot(\overline{\rho}\tilde{\mathbf{v}}\tilde{Y}_{i})=\nabla\cdot(\overline{\mathbf{j}_{i}}-\mathbf{b}_{i})+\dot{W}_{i} \end{cases}$$



Filtered constitutive equations

 $\overline{\mathbf{j}_{i}} \approx D_{i} \nabla \tilde{Y}_{i}, \ \overline{p} \approx \overline{\rho} R \tilde{T} \Sigma_{i} (\tilde{Y}_{i}/M_{i}), \ \overline{\mathbf{S}} \approx (\lambda + \frac{2}{3}\mu) (tr \tilde{\mathbf{D}}) \mathbf{I} + 2\mu \tilde{\mathbf{D}}_{D}, \ \overline{\mathbf{h}} = \overline{\kappa \nabla T} \approx \kappa \nabla \tilde{T}$

Filtered reaction rates and chemical kinetics

 $\overline{\dot{w}_{i}} = M_{i} \Sigma_{j=1}^{M} (P_{ij}'' - P_{ij}') \overline{\dot{w}_{j}} = M_{i} \Sigma_{j=1}^{M} (P_{ij}'' - P_{ij}') [\overline{k_{fj} \rho^{\Sigma_{i} P_{ij}'} \prod_{i=1}^{N} Y_{i}^{P_{ij}'} - k_{bj} \rho^{\Sigma_{i} P_{ij}''} \prod_{i=1}^{N} Y_{i}^{P_{ij}''}]$

Subgrid stress (**B**) and flux terms ($\mathbf{b}_{E} \& \mathbf{b}_{i}$) Definition: $\mathbf{B} = \overline{\rho}(\widetilde{\mathbf{v} \otimes \mathbf{v}} - \widetilde{\mathbf{v}} \otimes \widetilde{\mathbf{v}}), \ \mathbf{b}_{E} = \overline{\rho}(\widetilde{\mathbf{v} E} - \widetilde{\mathbf{v} E}), \ \mathbf{b}_{i} = \overline{\rho}(\widetilde{\mathbf{v} Y_{i}} - \widetilde{\mathbf{v} Y_{i}})$ Model: $\mathbf{B} = -2\mu_{k}\widetilde{\mathbf{D}}, \ \mathbf{b}_{E} = -2\frac{\mu_{k}}{Pr_{t}}\nabla E_{i}, \ \mathbf{b}_{i} = -2\frac{\mu_{k}}{Sc_{i}}\nabla \widetilde{Y}_{i}; \ \mu_{k} = \widetilde{\rho}c_{k}\Delta k^{1/2}$ $OEEVM/LDKM: \ \partial_{t}(\overline{\rho}k) + \nabla \cdot (\overline{\rho}\widetilde{\mathbf{v}}k) = -\mathbf{B}\cdot\widetilde{\mathbf{D}} + \nabla \cdot (\mu_{k}\nabla k) - \overline{\rho}c_{\epsilon}k^{3/2}/\Delta$



LES Combustion Models: Overview

Flame usually thinner than the LES grid resolution ($\delta_u < \Delta$) The filtered reaction rate $\overline{\dot{w}_j}$ is highly non-linear \Rightarrow large local variations Turbulence chemistry interactions (TCI) very important

Specific modeling (of either equation set and/or terms) required

c/z equation flamelet models

Propagation based or filtering based (e.g. Veynante et al, Weller et al., ...)

G/z equation flamelet models

Interface tracking of the flame front (e.g., Pitch et al)

Ξ, Σ, S_u

Thickened Flame Model (TFM) + reduced chemistry

Artificially thicken the flame to fit on the grid (e.g., O'Rourke & Bracco, Collin et al)

EDC or PaSR models + reduced chemistry

Eddy Dissipation Concept or Partially Stirred Reactor subgrid TCI models (e.g., Fureby et al)

Transported & presumed PDF models + reduced chemistry Probabilistic approach using subgrid PDF (e.g., Pope, Givi *et al*)

Linear Eddy Models (LEM) + reduced chemistry

1D sub-models for reaction-mixing in each LES cell (Menon et al, Kerstein et al)



LES Combustion Models: EDC/PaSR

Multi-scale model based on the assumption that reactions take place on the smallest *fine structures* (*) embedded in the *surroundings* (⁰)

Subgrid balance equations $\begin{cases} \overline{\rho}(Y_{i}^{*} - \tilde{Y}_{i}) = (1 - \gamma^{*})\tau^{*} \dot{w}_{i} (\overline{\rho}, Y_{i}^{*}, T^{*}) \\ \overline{\rho} \sum_{i=1}^{N} (Y_{i}^{*} h_{i}^{*} - \tilde{Y}_{i} \tilde{h}_{i}) = (1 - \gamma^{*})\tau^{*} \sum_{i=1}^{N} h_{i,f}^{\theta} \dot{w}_{i} (\overline{\rho}, Y_{i}^{*}, T^{*}) \end{cases}$

Need to determine τ^* and γ^*

EDC model

- Cascade process (v*, τ *)
- K41 consistent

•
$$v^* \approx v_K$$
, $\ell^* \approx 2\ell_K$, $\tau^* \approx 2\tau_K$

- Reaction space: tubes/sheets at high T
- $\gamma^* \approx \chi (v^*/v')^2 \approx \chi (\mu/\mu_k)^{3/4}$

PaSR model

- K41 hypothesis
- τ * based on [τ_{K} , τ'] τ * \approx ($\tau_{K}\tau'$)^{1/2}, τ' = Δ /v'
- Reaction space: tubes/sheets at high T
- $\gamma^* \approx \tau_c / (\tau_c + \tau^*)$



 $\partial_t (\overline{\rho} \tilde{Y}_i) + \nabla \cdot (\overline{\rho} \tilde{v} \tilde{Y}_i) = \nabla \cdot ((D_i + \mu_k / Sc_k) \nabla Y_i) + \gamma^* M_i P_{ij} \dot{w}_j (Y_i^*, T^*)$



Chemical Kinetics

Describing chemical kinetics (with sufficient accuracy and degree of detail) is very difficult due to the complexity of the reaction mechanisms Hydrogen: H₂-air; 8 species, 38 reactions (O'Conaire *et al* 2004) Methane: CH₄-air; 53 species, 325 reactions (GRI3.0) N-Heptane: C₇H₁₆-air; 561 species, 2539 reactions (Lu & Law 2008) Jet-A: C₁₂H₂₃-air; 18 species, 46 reactions (Yungster & Breisacher 2005)

Design of 'detailed reaction mechanisms'

- Identify all possible reactions \Rightarrow Reaction mechanism
- Collision theory $\dot{w} = \sigma A (\frac{8k_BT}{\pi u_{AB}})^{1/2} c_{\alpha} c_{\beta} e^{-E_A/RT}$
- Experimental data fits
 - Flame speed measurements
 - Ignition delay measurements Laser
 - Flow reaction measurements
- Simulations of measurements

How much chemistry do we need? and for which purpose?





Chemical Kinetics cont'd

http://www.me.berkeley.edu/gri_mech/version30/files30/grimech3...

1 See REALMENT file at anonymour PTP site unix.eri.com, directory gri) 1 WorldwideMeb home page http://www.me.berkeley.edu/grimech/ or 1 through http://www.grilorg , under "Basic Research",	http://www.ane.berkeley.cdu/	gri_mexh/version30/ülle/30/grimech3		
I for additional information, contacts, and disclaimer ELEMENTS	And a second			
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REACTIONS	TROE/ .5620 91.00 5036.00 0552.00/	LCW / 1.990E+41 -7.080 5685.00/	H02+CH3<=>02+CH4	1.000E+12 .000 .00
20+Hs=>02+Ht 1,200E+17 -1.000 .00	H2/2.00/ H20/6.00/ CH4/2.00/ CO/1.50/ CO2/2.00/ C2H6/3.00/ NR/ .70/	TROE/ ,8422 125.00 ,2219.00 5882.00 /	B02+CH3<=>OH+CH30	3.7802+13 .000 .00
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H2/2.00/ H20/6.00/ CH4/2.00/ Cg/1.50/ CG2/2.00/ C2H6/3.00/ NR/ .70/	H+CH3(+M)<=5CH4(+M) 13,90E+15534 536.00	H+C286C+0C285+82 1,1502+08 1,900	0+02<=>0+02	5.800E+13 .000 576.00
0+82<=>8+08 3.8708+04 2.700 5260.09	TROE/ .7830 74.00 2941.00 6954.00 /	H+HOCO<=>CH2(S)+CO 1.000E+14 .000	C+CH2<=>H+C2H	5.0002413 .000 .00
0+H02<=>0H+02 2.000E+13 .000 .00	H2/2.60/ H20/6.60/ CH4/3.60/ CO/1.50/ CO2/2.00/ C2H6/3.00/ AR/ .70/	H+CH2COM=>HCCD+H2 5.000E+13 .000	C+CH3<>H+C2H2	5.000E+13 .000 .00
0+8202<=>08+802 9.8302+05 2.000 4000.00	H+CH4<>>CH3+H2 6.600E+08 1.620 10840.00	H+CH2CDC=>CH3+CD 1.130E+13 .000	CH+02<=>0+800	5.710E+13 ,000 ,00 1.040E+15 000 1110.00
0+CH2<=>H+BC0 8.009E+13 .000 .00	H+HC0[+M] ===CH20[+M] 1.099E+12 .480 -260.09	H2+C0(+M)<23CH201+M1 4,300E+07 1,508	CH+H2OC>H+CH2O	5,710£+12 .000 -755.00
0+CH2(E)<>>H2+CD 1.500E+13 .000 .00	TROE/ .7824 271.00 2755.00 5570.00 /	LOW / 5.070E+27 -3.420 84350.00/	CH+CH2<=>H+C2H2	4.000E+13 .000 .00
0+CH2(S)<=>H+BCD 1.500E+13 .000 .00	H2/2.00/ H20/6.00/ CH4/2.00/ CG/1.50/ CO2/2.00/ C2H6/3.00/ NR/ .70/	TROE/ .9320 197.00 1540.00 10300.00 /	CH+CH3<=>H+C2H3	3.000E+13 .000 .00
0+CR1000R+CR20 5.060E+13 .000 .00	H#RC0cm5H2+C0 7.3408+13 .000 .00	H2/2.80/ H20/6.80/ CH4/2.80/ CO/1.50/ CG2/2.00/ C2H6/3.90/ A	CB+CB4<=>B+C2B4	5.009E+13 .090 .09
0+CD/+N)<>>CO2/+N) 1.800E+10 .000 2385.00	H+CH2O(+M)<>>CH2OH(+M) 5.400E+11 .454 3600.00	2.1602+08 1.510 200/401/202/401 7.4002+13 - 370	LOW / 2.690E+28 -3.740 193	5.00/
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0+800e=508+00 3,008E+13 .000 .00	H+CH2O(+M)<>>CH3O(+M) 5.400E+11 .454 2600.00	H2/2.00/ H2O/6.00/ CH4/2.00/ CO/1.50/ CO2/2.00/ C2H6/3.00/ #	CH+C02<=>HC0+C0	1,900E+14 .000 15792.00
0+CH2CK=>0H+HC0 3,900E+13 ,000 3540,00	LCW / 2.2000+30 -4.800 5560.00/	208<************************************	CH496COC+>CD4C2H2	5.000E+13 .000 .00
04CH2080=>0H4CH20 1.000E+13 .000 .00	1805/ ,1580 94.00 1555.00 4200.00 / 12/2 00/ 420/5 00/ rm2/2 00/ r0/1 50/ r02/2 00/ r245/1 00/	DUPLICATE	CH2+02=>08+8+CO	5.000E+12 .000 1500.00
0+CH3O<=>0H+CH2O 1.000E+13 .000 .00	H+CH20<=>HC0+H2 5.740E+07 1.900 2742.00	OH+H202<=>H02+H20 2.000E+12 .000	CH2+H2<=>H+CH3	5.000E+05 2.000 7230.00
O+CH108<=>OH+CH2OH 3.880E+05 2.500 3100.00	H+CH2OH(+H)<=>CH3OH(+H) 1.0552+12 .500 86.00	DUPLICATE	2CH2<=>H2+C2H2	1.500E+15 .000 11944.00
04/280/00/04/00 5.000/04 3.000 .00	LOW / 4.360E+31 -4.650 5080.00/	OH+H202<=5R02+H20 1.700E+18 .000	CH2+CH3<=>3(H)	4,000E+13 ,000 ,00 2,460E+05 2,000 8270,00
0+C2H2<>>H+HCC0 1.350E+07 2.000 1900.00	TROE/ .600 100.00 90900.0 10000.0 /	OH+C<=>H+CO 5,000E+13 ,000	(SI2+CD(+M)<>>CH2CD(+M)	8,1002+11 .500 4510.00
0+C2E2<=>0H+C2E 4.600E+19 =1.410 28950.00	E+CE208<>>H2+CE20 2,0005+13 ,000 ,00	08+CEc=>8+800 3.000E+13 .000	LCW / 2.690E+31 =5.110 705	5100/
0+C2H2<=>C0+CH2 5.940E+05 2.000 1900.00	H+CH20R<=>OH+CH3 1.650E+11 ,650 -294.00	CH+CH2<=>H+CH2C 2.000E+13 .000	TROE/ .5907 275.00 1226.00	5185.00 /
0+C2R3<=>R+CR2C0 3.000E+13 .000 .00 0+C2R3<=>CR34EC0 1.250E+07 1.830 .250.00	H+CH20B<4>CH2(5)+420 3,280E+13090 610.00	OB+CH2<⇒CH+H2O 1.130E+07 2.000	H2/2.00/ H2O/6.00/ CH4/2.00/ CO/1.50/	CO2/2.00/ C2H6/3.00/ AR/ .70/
0+C2B5<=>CH3+CH2O 2.240E+13 .000 .00	H+CHS0(+N)<=>CH30H(+N) 2.430E+12 .515 50.00	OB+CH3(+M)<=>CH3OH(+M) 2,790E+18 -1,430	QH2(5)+N2<>CH2+H2	1.5000+13 .000 600.00
0+C2H6<=>0H+C2H5 8.980E+07 1.920 5690.00	TROE/ .700 100.00 90000.0 10000.00 /	LEW / 4.000E+36 -5.920 3140.00/	CH2(8)+AR<=>CH2+AR	9.000E+12 .000 600.00
048000cm>H+200 1.000E+14 .000 .00	H2/2.80/ H2O/6.80/ CH4/2.80/ CO/1.50/ CO2/2.00/ C2H6/3.00/	TROE/ .4120 195.0 5900.00 6394.00/	CH2(5)+O2<~>H+OH+CO	2.809E+13 .000 .09
0+CH2C000508+RCC0 1,000E+13 ,000 8000,00 (4/02/02/02/02/02) 1,750E417 ,000 1350.00	H+CH3O<=>H+CH2CH 4,150E+07 1,630 1924.00	H2/2.00/ H2O/6.00/ CH4/2.00/ CO/1.50/ CO2/2.00/ C2H6/3.00/	CH2(S)+02<=>CD+H20	1.200E+13 .000 .00 2.000E+12 .000 .00
02+CD<=>0+CD2 2.500E+12 .000 47800.00	H+CH30c=>H2+CH20 2.000E+13 .000 .00 H+CH30c=>OH4CH3 1.500E+13 .000 .00	OH+CH3<>>CH2(S)+H20 6,440E+17 =1,340	CH2(8)+H2O(+M)<=>CH3OH(+M)	4.820E+17 -1.160 1145.00
02+CH2O<=>H02+HC0 1.000E+14 .000 40000.00	H+CH3O<=>CH2(S)+H2O 2.520E+14230 1070.09	OB+CH4<=>CH3+H20 1.000E+08 1.608	LOW / 1.888E+38 -6.350 594	9.007
H+02+09=>H02+09 2,800E+18 -,850 .09	H+CH308<=>CH208+H2 1.700E+07 2.100 4870.00	OH+COC=>H+CO2 4.760E+07 1.228	TROE/ .6027 208.00 3922.00	10180.0 /
8+202cm5802+02 2.0805+19 a1.240 .00	8+CB308<>CB30+62 \$,2002+05 2.103 4870.00	08+800<=>820+00 5:009E+13 .000	(92/2.00/ 020/5.00/ CH4/2.00/ CD/1.30/	3.0005+13 .000 .00
8+02+820<=>802+820 11.26E+18 <.760 .00	E+C2E(+E)<=>C2E2(+E) 1,000E+17 -1,000 .00 1768 / 3,7582+33 -4,000 1000.007	OH+CH20H<⇒H2O+CH2O 5,000E+12 ,000	CH2151+CH3<=>H+C2H4	1,2008+13 .000 -570.09
H+02+N2<=>H02+N2 2.500E+19 -1.240 .09	TROE/ .6464 132.00 1315.00 5566.00 /	OH+CH3D<=>H2O+CH2O 5.000E+12 .000	CH2(E)+CH4<>>2CH3	1.6002413 .000 -570.00
N+02+AR<=>R02+AR 7.000E+17800 .00	H2/2.00/ H20/6.00/ CH4/2.00/ CO/1.50/ CO2/2.00/ C2H6/3.00/ AR/ .70/	OB+CH3OB<=>CH2OB+H2O 1.440E+05 2.000	CH2(5)+COX=>CH2+CO	9.000E+12 .000 .00
H+02cHS0+QH 2,650E+16 -,6707 17041.00	H+C2H2(+M)<=>C2H3(+M) 5.600E+12 .000 2400.00	OH+CH3OH<⇒CH3O+H2O 6.300E+06 2.000	CH2(8)+C02<=>CH2+C02	(.000E+12 .000 .00 1.000E+12 .000 .00
	LXW / 3.800E+40 -7.270 7229.00/	08+C28<=>8+8CE0 2.0000+13 .000 /084C282<=>8460200 2.1808-01 4.500	CH2431+C2H6 <t>CH3+C2H5</t>	4.0002413 .000 ~550.00
and a	1805/ .7507 98.30 1302.00 4167.00 7 P2/2.60/ 420/5.00/ F92/2.60/ F0/1.50/ F02/2.00/ F246/3.00/ AB/ .70/	GB+C2H2<=>B+BCCGH 51040E+05 2.300	CH3+02<=>0+CH30	3.560E+13 .000 80480.00
	H+C2H3(+M)<=>C2H4(+M) 6.080E+12 .270 280.00	OH+C2H2<=>C2H+H2O 3.370E+07 2.000	CH3+02<=>OH+CH20	2.310E+12 .000 20315.00
	LXM / 1.400E+30 -3.060 3320.00/	OE+⊂2H2<=>CH3+CO \$.839E+0\$ \$.000	CH3+H202<=>H02+CH4	2.450E+04 2.470 5180.00
		OH+C2H3<>R2O+C2H2 5.000E+12 .000	100 / 3.400544) -7.030 274	6.//02+16 -1.180 654.00 2.60/
		OH+C2H6<=>C2H5+H20 3.540E+06 2.120	TROE/ .6190 73.20 1180.00 5	999.00 /
	Z un H	OE+CH2CO<⇒RCCO+B2O 7.500E+12 .008	H2/2.00/ H20/6.00/ CH4/2.00/ CO/1.50	CO2/2.00/ C286/3.00/ AR/ .70/
		2H02<=>02+H202 1.300E+11 .000	2CH3<=>H+C2H5	6.840E+12 .100 10600.00
		DOPLICATE	(%)3+8C0005C84+C0	2.5488+13 .000 .00
			CH3+CH3ORC+>CH2OH+CH4	3,000E+07 1,500 9940,00
			CH3+CH308<=>CH30+CR4	1.0002+07 1.500 9940.00
		B an B	CH3+C2H4<⇒C2H3+CH4	2.270E+05 2.000 9200.00
			CH3+C2R6<>C2R5+CR4 HC0+R2Ocra8+CD4R2O	5.1402+05 1.740 10450.00 1.5002+18 -1.000 17000.00
			And All and Al	
			2.4	11.01 10.71.02
				11-01-50 23.08

http://www.me.berkeley.edu/gri_mech/version30/text30.html



Reduced Chemical Kinetics

Example: Jet A is a kersone grade fuel with a carbon number distribution between 8 and 16.

Jet A can be assumed to consist of C_8H_{18} , $C_{10}H_{22}$, $C_{12}H_{22}$, $C_{12}H_{24}$, $C_{14}H_{26}$ and $C_{16}H_{28}$ with the average molecular formula $C_{12}H_{23}$.

Jet-A: C₁₂H₂₃-air; 18 species, 46 reactions (Yungster & Breisacher 2005)

Reduction technique of Meredith & Black based on SQP to simulate a setof continuously stirred tank reactors.Adiabatic Flame TemperatureOPContinuously stirred tank reactors.

- \Rightarrow 5 species and 2 reactions
- \Rightarrow Acceptable agreement for 0.3< ϕ <1.3

Reaction		A [m, kg, mol,	$s = n_T$	n _{C12H23}	no	
$C_{12}H_{23}+11.75O_2 \rightarrow 12CO+11.5H_2O$ CO+0.5O ₂ \rightarrow CO ₂			$\begin{array}{r} 1.04{\cdot}10^9 \\ 4.04{\cdot}10^8 \\ 7.14{\cdot}10^{13} \end{array}$	0	1.0	0.5 0.5 0.5
				0		
$O_2 + N_2 \rightarrow 2NO$		-0.5				
_		Table 2. Schn	ndt numbers			1
	also and a second s			II O	N	
Specie	C12H23	0 ₂ C(CO_2	H ₂ U	112	



Yungster & Breisacher AIAA 2005-4210 Meredith & Black AIAA 2006-1168



Validation: Low Swirl Burner



Nogenmyr et al.; 2008, AIAA 2008-0513 Nogenmyr et al.; 2009, Comb. Flame. **156**, p 25



Detailed Physics: Planar Flame in HIT

Need to understand how different LES combustion models capture the at least a canonical flame. DNS & LES of a box of size 18 mm.



CESAR Aero GT Engine Combustor

Fedina et al AIAA 2011-0785



The HYSHOT Combustor

A detailed CFD study of the flight experiments must incorporate also the HEG nozzle and test section. **Combine RANS and LES**.

Stand alone RANS in HEG nozzle (DLR)



Axisymmetric RANS model with 20,000 cells

RANS in HEG test section + RANS/LES in combustor



2D planar RANS (15,000 cells) of the flow in the entire HEG test section $T_0=1459 \text{ K}$. 3D RANS (1 Mcells) and 3D LES (12.5/25.0 Mcells) of the flow in the combustor

Karl *et al*, 2006, AIAA 2006-8041 Fureby *et al*, 2010, Proc. Comb. Inst.



Large Scale Global Flow Features





Fureby et al, 2010, Proc. Comb. Inst.

Validation & Combustion Dynamics



Fureby et al, 2010, Proc. Comb. Inst.





Bow shock causes BL to separate and H₂ to leak into downstream BL region

H₂ rich jet shear layer undergoes KH instabilities

H₂ rich side arms (S-vortices) and spanwise rollers (Q-vortices) are formed

The H₂ rich S and Q-vortices are transported downstream under intense shear during which H₂ mix macroscopically with hot air. S and Q vortices merge to Ω -vortices.

Stretching increases the interfacial area and steepens the concentration gradients whilst enhancing the diffusive micromixing.

Self-ignition takes place after micromixing is complete.





Fureby et al, 2010, Proc. Comb. Inst.

Can Modeling of Reactive Flows Describe Reality?

YES and NO.

The computational framework (models, methods and codes) is now available. The computational infrastructure is somewhat limited but is rapidly becoming available.

Need to develop a better understanding of the chemistry and how to model certain processes (TCI, ...) (sub-models). Importance?

How does turbulence influence ... and vice versa.

- Scalar mixing
- Combustion chemistry
- Conjugate heat transfer
- Acoustics

Need to better work with the experimentalists.

Need to use our skills to demonstrate to companies, funding agencies, etc. that engineering problems can be computed.

Things evolve ... Plasma assisted combustion ...

